# On a modification of GLS stabilized FEM for solving incompressible viscous flows

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### SUMMARY

We deal with 2D flows of incompressible viscous fluids with high Reynolds numbers. Galerkin Least Squares technique of stabilization of the finite element method is studied and its modification is described. We present a number of numerical results obtained by the developed method, showing its contribution to solving flows with high Reynolds numbers. Several recommendations and remarks are included. We are interested in positive as well as negative aspects of stabilization, which cannot be divorced. Copyright © 2005 John Wiley & Sons, Ltd.

KEY WORDS: FEM; incompressible flow; stabilization; GLS; semiGLS

#### 1. INTRODUCTION

The idea of stabilizing the FEM is not quite new in comparison to the history of FEM itself. Many researchers have been involved in this area and have already presented many techniques and results. Let us briefly review several publications related to the approach that we present in this paper. Hughes *et al.* [1] presented the stable Petrov–Galerkin formulation of the Stokes problem in 1986. Douglas and Wang [2] introduced another stabilized method for the Stokes problem in 1988. In the same year; Hughes *et al.* [3] presented SUPG and GLS stabilized finite element methods for the advective–diffusive equation. Their ideas were extended to the Navier–Stokes equations and completed by Franca and Frey [4], Franca and

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Hughes [5], Franca and Madureira [6], and Franca *et al.* [7]. In 1991, Tezduyar presented pressure-stabilizing/Petrov–Galerkin (PSPG) method, another stabilization technique for incompressible viscous fluids [8]. Work of Lube and his co-workers (e.g. Reference [9]) can be mentioned as a recent research in the stabilization of the FEM for fluid dynamics.

Mainly the works of Franca and Hughes provide the theoretical basis for the presented research. We modify the Galerkin Least-Squares (GLS) method introduced in Reference [3] and solve flows with markably higher Reynolds numbers than without stabilization. We present several results of numerical experiments which show the impact of this stabilization on the solution and people who use stabilization techniques should be aware of it.

# 2. MODEL PROBLEM

Let  $\Omega$  be an open bounded domain in  $\mathbb{R}^2$  filled with an incompressible viscous fluid, and let  $\Gamma$  be its boundary. Isothermal flow of such fluid is governed by the following Navier–Stokes system of partial differential equations (nonconservative form)

• Unsteady flow

$$\frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla)\mathbf{u} - v\Delta \mathbf{u} + \nabla p = \mathbf{f} \quad \text{in } \Omega \times [0, T]$$
(1)

$$\nabla \cdot \mathbf{u} = 0 \quad \text{in } \Omega \times [0, T] \tag{2}$$

$$\mathbf{u} = \mathbf{g} \quad \text{on } \Gamma_g \times [0, T] \tag{3}$$

$$-v(\nabla \mathbf{u})\mathbf{n} + p\mathbf{n} = \mathbf{0} \quad \text{on } \Gamma_h \times [0, T]$$
(4)

$$\mathbf{u} = \mathbf{u}_0 \quad \text{in } \Omega, \ t = 0 \tag{5}$$

• Steady flow

$$(\mathbf{u} \cdot \nabla)\mathbf{u} - v\Delta \mathbf{u} + \nabla p = \mathbf{f} \quad \text{in } \Omega \tag{6}$$

$$\nabla \cdot \mathbf{u} = 0 \quad \text{in } \Omega \tag{7}$$

$$\mathbf{u} = \mathbf{g} \quad \text{on } \Gamma_{q}$$
 (8)

$$-v(\nabla \mathbf{u})\mathbf{n} + p\mathbf{n} = \mathbf{0} \quad \text{on } \Gamma_h \tag{9}$$

where

- *t* denotes time variable,
- $\mathbf{u} = (u_1, u_2)^{\mathrm{T}}$  denotes the vector of flow velocity,
- *p* denotes the pressure divided by the density,
- v denotes the kinematic viscosity of the fluid supposed to be constant,
- f denotes the density of volume forces per mass unit,
- $\Gamma_g$  and  $\Gamma_h$  are two subsets of  $\Gamma$  satisfying  $\overline{\Gamma} = \overline{\Gamma}_g \cup \overline{\Gamma}_h$ ,  $\mu_{\mathbb{R}^1}(\Gamma_g \cap \Gamma_h) = 0$ ,
- **n** denotes an outer normal vector to the boundary  $\Gamma$  with unit length,
- **g** is a given function satisfying  $\int_{\Gamma} \mathbf{g} \cdot \mathbf{n} \, d\Gamma = 0$  in the case of  $\Gamma = \Gamma_g$ ,
- $\mathbf{u}_0$  is a given flow field satisfying  $\nabla \cdot \mathbf{u}_0 = 0$ .

# 3. APPROXIMATION OF THE PROBLEM BY FEM

First we derive the weak formulation of the Navier–Stokes equations (1)-(5) in the manner of mixed methods, i.e. usage of different function spaces of test functions for the momentum equation and for the continuity equation (cf. Reference [10]). Let us define vector function spaces  $V_q$  and V by

$$V_{g} = \{ \mathbf{v} = (v_{1}, v_{2})^{\mathrm{T}} \mid \mathbf{v} \in [H^{1}(\Omega)]^{2}; \mathbf{Tr} \ v_{i} = g_{i}, i = 1, 2 \}$$
$$V = \{ \mathbf{v} = (v_{1}, v_{2})^{\mathrm{T}} \mid \mathbf{v} \in [H_{0}^{1}(\Omega)]^{2} \}$$

where  $H^1(\Omega)$  and  $H^1_0(\Omega)$  are the usual Sobolev spaces. Let additionally  $L_2(\Omega)$  be the space of square integrable functions on  $\Omega$ , and let  $L_2(\Omega)/\mathbb{R}$  be the space of functions in  $L_2(\Omega)$  ignoring an additive constant.

Then the weak unsteady Navier–Stokes problem consists of finding  $\mathbf{u}(t) = (u_1(t), u_2(t))^T \in V_g$  and  $p(t) \in L_2(\Omega)/\mathbb{R}$  satisfying for any  $t \in [0, T]$ 

$$\int_{\Omega} \frac{\partial \mathbf{u}}{\partial t} \cdot \mathbf{v} \, \mathrm{d}\Omega + \int_{\Omega} (\mathbf{u} \cdot \nabla) \mathbf{u} \cdot \mathbf{v} \, \mathrm{d}\Omega + \nu \int_{\Omega} \nabla \mathbf{u} : \nabla \mathbf{v} \, \mathrm{d}\Omega - \int_{\Omega} p \nabla \cdot \mathbf{v} \, \mathrm{d}\Omega = \int_{\Omega} \mathbf{f} \cdot \mathbf{v} \, \mathrm{d}\Omega \tag{10}$$

$$\int_{\Omega} \psi \nabla \cdot \mathbf{u} \, \mathrm{d}\Omega = 0 \tag{11}$$

$$\mathbf{u} - \mathbf{u}_g \in V \tag{12}$$

for all  $\mathbf{v} \in V$  and  $\psi \in L^2(\Omega)$ , where  $\mathbf{u}_g \in V_g$  is a representation of the Dirichlet boundary condition  $\mathbf{g}$  in (3). We denote

$$\nabla \mathbf{u} : \nabla \mathbf{v} = \frac{\partial u_1}{\partial x_1} \frac{\partial v_1}{\partial x_1} + \frac{\partial u_1}{\partial x_2} \frac{\partial v_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \frac{\partial v_2}{\partial x_1} + \frac{\partial u_2}{\partial x_2} \frac{\partial v_2}{\partial x_2}$$

For the steady problem (6)–(9), we obtain similar formulation as (10)–(12) without the term with the time derivative in (10) and dependence on time.

Let us divide the domain  $\Omega$  into N elements  $T_K$  of a triangulation  $\mathscr{T}_h$  of shape regular family, such that  $\bigcup_{K=1}^N \overline{T}_K = \overline{\Omega}$ ,  $\mu_{\mathbb{R}^2}(T_K \cap T_L) = 0$ ,  $K \neq L$ . Let  $h_K$  mean the largest distance in element  $T_K$ .

In what follows, we consider Hood–Taylor finite elements  $P_2P_1$  and/or  $Q_2Q_1$ , which satisfy Babuška–Brezzi stability condition (*inf–sup* condition) (cf. Reference [11]). Their application leads to the final approximation on the domain  $\Omega$  satisfying  $\mathbf{u}_h \in V_{gh}$  and  $p_h \in Q_h$  where

$$V_{gh} = \{ \mathbf{v}_h = (v_{h_1}, v_{h_2})^{\mathrm{T}} \in [\mathscr{C}(\bar{\Omega})]^2; \ v_{h_i} \mid_{T_K} \in R_2(\overline{T_K}), \ K = 1, \dots, N, \ i = 1, 2, \ \mathbf{v}_h = \mathbf{g} \}$$

in nodes on  $\Gamma_a$ 

$$Q_h = \{ \psi_h \in \mathscr{C}(\overline{\Omega}); \ \psi_h \mid_{T_K} \in R_1(\overline{T_K}), \ K = 1, \dots, N \}$$

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where

$$R_m(\overline{T_K}) = \begin{cases} P_m(\overline{T_K}) & \text{if } T_K \text{ is a triangle} \\ Q_m(\overline{T_K}) & \text{if } T_K \text{ is a quadrilateral} \end{cases}$$

and  $\mathscr{C}(\bar{\Omega})$  denotes the space of continuous functions on  $\bar{\Omega}$ . Additionally, we introduce the space

$$V_h = \{ \mathbf{v}_h = (v_{h_1}, v_{h_2})^{\mathrm{T}} \in [\mathscr{C}(\bar{\Omega})]^2; \ v_{h_i} \mid_{T_K} \in R_2(\overline{T_K}), \ K = 1, \dots, N, \ i = 1, 2, \ \mathbf{v}_h = \mathbf{0}$$

in nodes on  $\Gamma_a$ 

Since these function spaces satisfy  $V_{gh} \subset V_g$ ,  $V_h \subset V$ , and  $Q_h \subset L_2(\Omega)/\mathbb{R}$ , we can introduce *semidiscrete unsteady Navier–Stokes problem*:

Find  $\mathbf{u}_h(t) \in V_{gh}$ ,  $t \in [0, T]$  and  $p_h(t) \in Q_h$ ,  $t \in [0, T]$  satisfying for any  $t \in [0, T]$ 

$$\int_{\Omega} \frac{\partial \mathbf{u}_{h}}{\partial t} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega + \int_{\Omega} (\mathbf{u}_{h} \cdot \nabla) \mathbf{u}_{h} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega + v \int_{\Omega} \nabla \mathbf{u}_{h} : \nabla \mathbf{v}_{h} \, \mathrm{d}\Omega$$
$$- \int_{\Omega} p_{h} \nabla \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega = \int_{\Omega} \mathbf{f} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega, \quad \forall \mathbf{v}_{h} \in V_{h}$$
(13)

$$\int_{\Omega} \psi_h \nabla \cdot \mathbf{u}_h \, \mathrm{d}\Omega = 0, \quad \forall \psi_h \in Q_h \tag{14}$$

$$\mathbf{u}_h - \mathbf{u}_{gh} \in V_h \tag{15}$$

#### 4. STABILIZED FORMULATION

Following ideas of Hughes and Franca [3, 6], we apply the GLS stabilizing technique with two modifications:

- 1. Stabilization of the continuity equation is not considered. For this reason, we call the technique *semiGLS* (abbreviated sGLS). Since most problems are caused by the advective term, we find stabilization of it as most important. We performed several experiments with stabilization of the continuity equation applied, but the results were disastrous.
- 2. The algorithm is derived for the Navier–Stokes equations with Laplacian in the diffusive term instead of whole symmetric part of the velocity gradient. This formulation makes the derivations much simpler, therefore less mistakes can be made and stay undetected.

Applying stabilization to the momentum equation (13) and adding the continuity equation (14), we introduce the stabilized problem:

Find  $\mathbf{u}_h(t) \in V_{gh}$ ,  $t \in [0, T]$  and  $p_h(t) \in Q_h$ ,  $t \in [0, T]$  satisfying for any  $t \in [0, T]$ 

$$B_{\text{sGLS}}(\mathbf{u}_h, p_h; \mathbf{v}_h, \psi_h) = L_{\text{sGLS}}(\mathbf{v}_h, \psi_h), \quad \forall \mathbf{v}_h \in V_h, \quad \forall \psi_h \in Q_h$$
(16)

$$\mathbf{u}_h - \mathbf{u}_{gh} \in V_h \tag{17}$$

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where

$$B_{\text{sGLS}}(\mathbf{u}_{h}, p_{h}; \mathbf{v}_{h}, \psi_{h}) \equiv \int_{\Omega} \frac{\partial \mathbf{u}_{h}}{\partial t} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega + \int_{\Omega} (\mathbf{u}_{h} \cdot \nabla) \mathbf{u}_{h} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega$$
$$+ v \int_{\Omega} \nabla \mathbf{u}_{h} : \nabla \mathbf{v}_{h} \, \mathrm{d}\Omega - \int_{\Omega} p_{h} \nabla \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega + \int_{\Omega} \psi_{h} \nabla \cdot \mathbf{u}_{h} \, \mathrm{d}\Omega$$
$$+ \sum_{K=1}^{N} \int_{T_{K}} \left[ \frac{\partial \mathbf{u}_{h}}{\partial t} + (\mathbf{u}_{h} \cdot \nabla) \mathbf{u}_{h} - v \Delta \mathbf{u}_{h} + \nabla p_{h} \right]$$
$$\cdot \tau \left[ (\mathbf{u}_{h} \cdot \nabla) \mathbf{v}_{h} - v \Delta \mathbf{v}_{h} + \nabla \psi_{h} \right] \, \mathrm{d}\Omega$$
$$L_{\text{sGLS}}(\mathbf{v}_{h}, \psi_{h}) \equiv \int_{\Omega} \mathbf{f} \cdot \mathbf{v}_{h} \, \mathrm{d}\Omega + \sum_{K=1}^{N} \int_{T_{K}} \mathbf{f} \cdot \tau \left[ (\mathbf{u}_{h} \cdot \nabla) \mathbf{v}_{h} - v \Delta \mathbf{v}_{h} + \nabla \psi_{h} \right] \, \mathrm{d}\Omega$$

Here  $\tau$  is positive stabilization parameter. Modifying formulas from Reference [6], we compute it as

$$\tau = \frac{\zeta(Re_K(\mathbf{x}))}{\sqrt{\lambda_K} |\mathbf{u}(\mathbf{x})|_2}$$
(18)

where

$$Re_{K}(\mathbf{x}) = \frac{|\mathbf{u}(\mathbf{x})|_{2}}{4\sqrt{\lambda_{K}}\nu}$$
$$\xi(Re_{K}(\mathbf{x})) = \begin{cases} Re_{K}(\mathbf{x}), & 0 \leq Re_{K}(\mathbf{x}) < 1\\ 1, & Re_{K}(\mathbf{x}) \geq 1 \end{cases}$$
$$\lambda_{K} = \max_{0 \neq \mathbf{v} \in (R_{2}(T_{K})/\mathbb{R})^{2}} \frac{\|\Delta \mathbf{v}\|_{0, T_{K}}^{2}}{\|\nabla \mathbf{v}\|_{0, T_{K}}^{2}}$$
$$|\mathbf{u}(\mathbf{x})|_{2} = \left(\sum_{i=1}^{2} |u_{i}(\mathbf{x})|^{2}\right)^{1/2}$$

Parameter  $\lambda_K$  is computed for each element as the largest eigenvalue of the problem

$$(\Delta \mathbf{w}, \Delta \mathbf{v}) = \lambda_K (\nabla \mathbf{w}, \nabla \mathbf{v}) \quad \forall \mathbf{v} \in (R_2(T_K)/\mathbb{R})^2$$
(19)

This is done once, before entering the main computational loop of the Newton method, since  $\lambda_K$  is not a function of velocity and depends only on the computational mesh through space functions on element K.

Stabilized formulation for the steady problem (6)–(9) is derived analogously and is simpler. Let us investigate the dependence of  $\tau$  on local Reynolds number  $Re_K(\mathbf{x})$  given by (18). We can observe that  $Re_K(\mathbf{x})$  is a linear function of  $|\mathbf{u}(\mathbf{x})|_2$  for constant viscosity on element K, i.e.

$$Re_K(\mathbf{x}) = C_1 |\mathbf{u}(\mathbf{x})|_2 \tag{20}$$

where  $C_1 = 1/(4\sqrt{\lambda_K}v)$ .

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Figure 1. Plot of  $\tau(Re_K)$ .

Substituting (20) in (18) we get

$$\tau(Re_{\kappa}(\mathbf{x}),\mathbf{x}) = \begin{cases} C_2, & 0 \leq Re_{\kappa}(\mathbf{x}) < 1\\ \frac{C_3}{|\mathbf{u}(\mathbf{x})|_2} = \frac{C_2}{Re_{\kappa}(\mathbf{x})}, & Re_{\kappa}(\mathbf{x}) \geq 1 \end{cases}$$

where  $C_2 = 1/(4\lambda_K v)$  and  $C_3 = 1/\sqrt{\lambda_K}$ , cf. Figure 1.

# 5. IMPLEMENTATION OF THE METHOD

In our implementation, we employ the implicit Euler method (also known as the backward difference method) for time discretization of the problem (16)–(17), i.e. time derivative is substituted as

$$\frac{\partial \mathbf{u}_h}{\partial t} \approx \frac{\mathbf{u}_h^{n+1} - \mathbf{u}_h^n}{\vartheta}$$

where  $\vartheta$  denotes a constant time step. This leads to fully implicit method for finding  $\mathbf{u}_h$  in (n+1)th time layer. Resulting system of nonlinear equations is solved by the Newton method in each time layer, employing direct solver for solving the linearized system in each iteration.

# 6. NUMERICAL EXPERIMENTS

The method was tested on several problems for verification and to review its behaviour. Results obtained by the algorithm are marked as *semiGLS* algorithm results.

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Figure 2. Streamlines by the Newton method without stabilization (left) and by the semiGLS algorithm (right),  $Re = 10\,000$ , mesh  $32 \times 32$ .

## 6.1. Steady solution of lid driven cavity

Popular benchmark problem for testing numerical schemes is the 'lid driven cavity'. Computational domain is of square shape with unit length of side. Dirichlet boundary conditions are prescribed on the boundary: value of horizontal velocity is prescribed on the upper side, zero boundary conditions on the rest of the boundary representing a wall.

Many solutions of this problem were presented by various authors. Here are some representatives: in Reference [12], solutions for Reynolds numbers 1000, 3200, and 5000 obtained on nonuniform grid of approximately 8800 elements are presented; in Reference [9], result for Reynolds number 7500 on quasi-uniform mesh of  $96 \times 96$  elements is presented; solutions for Reynolds number 10 000 obtained by several methods on the mesh of  $64 \times 64$  elements are published in Reference [13], and in Reference [7], outstanding results for Reynolds number  $500\,000$  on the mesh of  $30 \times 30$  elements are presented.

Solution by the developed algorithm was performed on three uniform meshes of quadrilateral elements—of  $32 \times 32$  (1024) elements, of  $64 \times 64$  (4096) elements, and of  $128 \times 128$  (16 384) elements.

To observe the effect of stabilization, solutions obtained by the Newton method without stabilization together with solutions computed by the semiGLS algorithm for Reynolds number 10 000 on all three meshes are presented in Figures 2–4. Moreover, we can review the sensitivity to the fineness of the computational mesh in these figures.

We can observe, that streamlines are not encircled for the GLS method showing the loss of accuracy induced by the stabilization. This defect is decreasing with refining of the mesh.

We evaluate the effect of stabilization as the difference of solutions obtained with and without stabilization by

$$\delta_{\eta} = \sqrt{\frac{\sum_{i=1}^{n} (\eta_{\text{sGLS}_{i}} - \eta_{\text{Newton}_{i}})^{2}}{\sum_{i=1}^{n} \eta_{\text{Newton}_{i}}^{2}}} \times 100 \ (\%)$$
(21)

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Figure 3. Streamlines by the Newton method without stabilization (left) and by the semiGLS algorithm (right),  $Re = 10\,000$ , mesh  $64 \times 64$ .



Figure 4. Streamlines by the Newton method without stabilization (left) and by the semiGLS algorithm (right),  $Re = 10\,000$ , mesh  $128 \times 128$ .

where  $\eta$  represents in turn  $u_{h1}$ ,  $u_{h2}$  and  $p_h$ , n denotes number of nodes with  $\eta$  given,  $\eta_{sGLS}$  denotes the solution obtained by the semiGLS algorithm and  $\eta_{Newton}$  denotes the solution obtained by the Newton method without stabilization. Results are summarized in Table I.

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Table I. Differences between solutions obtained with and without stabilization.

Mesh	32×32	64×64	128×128
$\delta_{u_{h1}}$ (%)	41.69	39.07	21.42
$\delta_{u_{h2}}$ (%)	70.81	49.12	22.24
$\delta_{p_h}$ (%)	197.90	137.10	42.82



Figure 5. Streamlines (left) and pressure contours (right) by the semiGLS algorithm on the mesh  $128 \times 128$ ,  $Re = 100\,000$ .

Differences of solutions obtained by the semiGLS method from those obtained by the Newton method computed by (21) are big for the problem of cavity.

To extend results to higher Reynolds numbers, solution for  $Re = 100\,000$  on the mesh  $128 \times 128$  is presented in Figure 5.

Although the continuation method was applied to achieve higher Reynolds numbers, we detected limits of convergence of the Newton method for all three meshes. We observed, that on the mesh  $32 \times 32$ , we were not able to get results above  $Re \approx 28\,000$ , on the mesh  $64 \times 64$  above  $Re \approx 50\,000$ , and on the mesh  $128 \times 128$  above  $Re \approx 120\,000$ . For comparison, such limit was around  $Re \approx 12\,500$  on the mesh  $32 \times 32$  for the method without stabilization  $(\tau = 0)$ .

Another interesting effect was observed during the computations. Since it is known that stabilized methods are in general, sensitive to stabilization parameters, we tried to modify the computed parameter  $\tau$  by a quotient 0.7–1.5. This improved the convergence, and we were able to reach higher Reynolds numbers, e.g.  $Re = 70\,000$  on the mesh  $64 \times 64$  elements.



Figure 6. Geometry of the channel.



Figure 7. Streamlines in the channel by the Newton method without stabilization (left) and streamlines by the semiGLS algorithm (right), Re = 1000.

Table II. Differences between solutions obtained with and without stabilization.

Mesh	Channel (Figure 6)	
$\overline{\delta_{u_{h1}}}$ (%)	0.0718	
$\delta_{u_{h2}}$ (%) $\delta_{p_{L}}$ (%)	2.7202 0.5139	
Ph		

# 6.2. Steady solution of flow in channel with sudden extension of diameter

Steady flow in 2D channel with abruptly extended diameter (Figure 6) is another testing problem. This problem is complicated due to singularities of solution in the vicinity of nonconvex internal corners. The aspect of suitable mesh generation for such problems is studied in References [14, 15]. We compare solutions with and without semiGLS stabilization in this paper. Streamlines are presented in Figure 7 for Reynolds number 1000. For the symmetry of the problem, solution is found only on the upper half of the section. Differences between solutions computed by (21) are listed in Table II. Parabolic horizontal velocity distribution is prescribed on the inflow (left) part of the boundary, 'do nothing' condition is considered on the outflow (right) part of it, zero velocity is prescribed on the upper part of the boundary representing wall with 'no slip' and symmetry is considered on the lower part of it (see Reference [15] for details).

Additionally, we present streamlines, plots of velocities and pressure for Reynolds number 80 000 in Figures 8 and 9.



Figure 8. Streamlines (left) and plot of velocity  $u_{h1}$  (right) by the semiGLS algorithm,  $Re = 80\,000$ .



Figure 9. Plot of velocity  $u_{h2}$  (left) and pressure (right) by the semiGLS algorithm,  $Re = 80\,000$ .



Figure 10. Computational mesh for NACA 0012 problem, angle of incidence of 34°.

# 6.3. Unsteady solution of flow past NACA 0012 airfoil

Unsteady flow past NACA 0012 airfoil was investigated as a more practical application. Results of this problem for angle of incidence of  $34^{\circ}$  and Reynolds number 1000 obtained by the unconditionally stable projection FEM were presented by Guermond and Quartapelle in



Figure 11. Streamlines by the semiGLS algorithm (left) and by Guermond and Quartapelle [12] (right), t = 1.6 s, Re = 1000.



Figure 12. Pressure contours by the semiGLS algorithm (left) and by Guermond and Quartapelle [12] (right), t = 1.6 s, Re = 1000.

Reference [12]. In Figures 11–15, these results are compared to ours obtained by the semiGLS algorithm. Streamlines and pressure contours for the problem with Reynolds number 100 000 in several time layers are presented in Figures 16–19. The computational mesh is shown in Figure 10. It contains 6220 elements, 18 478 nodes, and 43 085 degrees of freedom. We consider zero initial condition, unit horizontal velocity on the left part of the boundary and 'do nothing' boundary condition on the rest of it. Time step sizes for simulation for Reynolds number 1000 are 0.01 s and for Reynolds number 100 000 are 0.005 s (see Reference [15] for details).

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Figure 13. Streamlines by the semiGLS algorithm (left) and by Guermond and Quartapelle [12] (right), t = 2.6 s, Re = 1000.



Figure 14. Streamlines by the semiGLS algorithm (left) and by Guermond and Quartapelle [12] (right), t = 3.6 s, Re = 1000.



Figure 15. Streamlines by the semiGLS algorithm (left) and by Guermond and Quartapelle [12] (right), t = 6.0 s, Re = 1000.

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Figure 16. Streamlines (left) and pressure contours (right) by the semiGLS algorithm, t = 1.6 s, Re = 100000.



Figure 17. Streamlines (left) and pressure contours (right) by the semiGLS algorithm, t = 2.6 s,  $Re = 100\,000$ .



Figure 18. Streamlines (left) and pressure contours (right) by the semiGLS algorithm, t = 3.6 s, Re = 100000.

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Figure 19. Streamlines (left) and pressure contours (right) by the semiGLS algorithm, t = 6.0 s, Re = 100000.

### 7. CONCLUSION

The motivation for using stabilization techniques comes from the desire of solving problems for higher Reynolds numbers.

We achieved our main goal—to develop an applicable algorithm based on the FEM equipped with stabilization. The algorithm has been verified by several numerical experiments and performed contributive results. More about the algorithm and other numerical results can be found in Reference [15].

On the problem of cavity for  $Re = 10\,000$  we show, that applying stabilization induces loss of accuracy. Distortion of solution could be significant for some problems (Figures 2–4 and Table I), but could also play negligible role for other problems (Figure 7 and Table II).

It was observed that presented stabilization technique leads to improvement of stability of the Newton method to approximately 'double' Reynolds number. Similar effect was observed for the refinement of the computational mesh with half element size.

This results in recommendation, that for solving flows for higher Reynolds numbers, applying of stabilization techniques should be efficiently combined with a suitable refinement. Stabilization should be applied carefully and reliable tool for error estimation for semiGLS technique is still desired.

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